

# Analysis of Beam Structures

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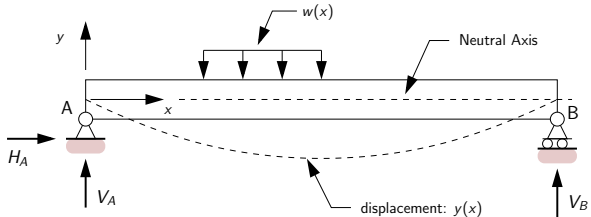
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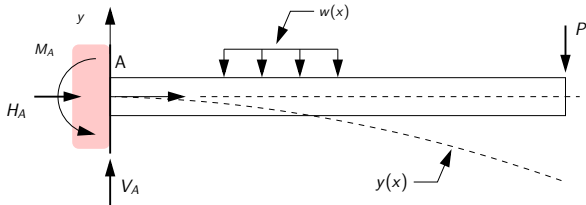
# Types of Beam Structure

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Simply Supported Beam:

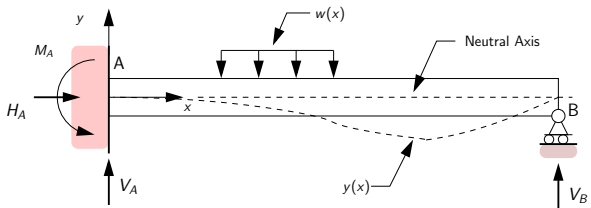


Cantilever:

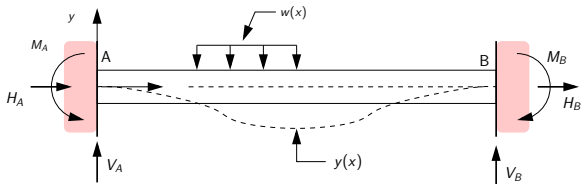


# Types of Beam Structures

Supported Cantilever:



Fixed-Fixed Beam Structure:



# Types of Beam Structures

## Boundary Conditions

### Simply Supported Beam

- $y(0) = y(L) = 0$ .

### Cantilever Beam

- $y(0) = 0, \frac{dy}{dx}|_{x=0} = 0$

### Supported Cantilever Beam

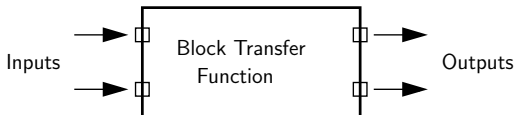
- $y(0) = y(L) = 0, \frac{dy}{dx}|_{x=0} = 0$

### Fixed-Fixed Beam

- $y(0) = y(L) = 0, \frac{dy}{dx}|_{x=0} = \frac{dy}{dx}|_{x=L} = 0$

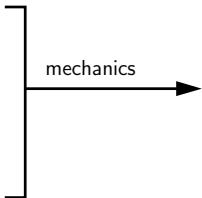
# Basic Questions

**Q1.** What is the relationship between inputs and outputs?



## Inputs

Applied loads ( $P$  and  $w$ )  
Boundary conditions  
Beam geometry ( $L$  and  $I$ )  
Material Properties ( $E$ )



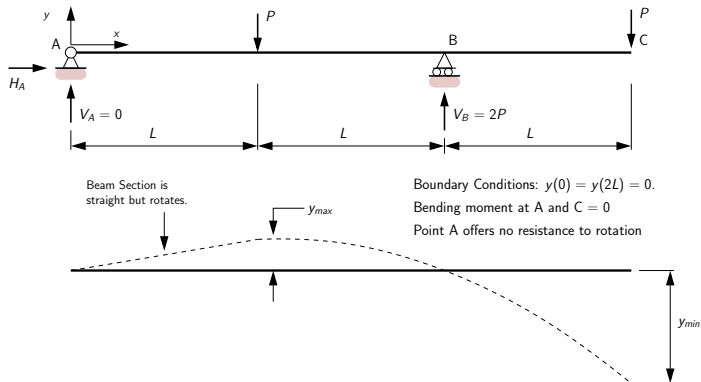
## Outputs

Shear force –  $V(x)$   
Bending moment –  $M(x)$   
Axial force –  $N(x)$   
Displacement –  $y(x)$   
Rotation –  $\theta(x) = \left[ \frac{dy}{dx} \right]$

Decisions will be based on estimates of outputs.

# Basic Questions

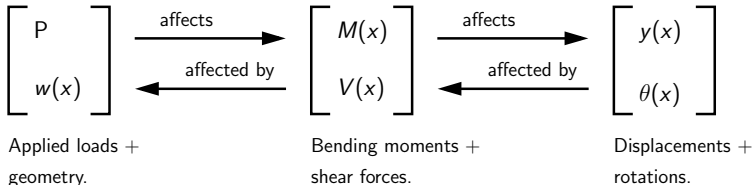
Typical problem: Given input parameters, compute  $y(x)$ , find location and magnitude of  $y_{min}$  and  $y_{max}$ .



For simple problems, can rely on intuition. Otherwise, need **math** and **mechanics**.

# Basic Questions

**Q2.** What is the relationship among the outputs? Are they dependent?



We will need to work with a chain of dependencies.

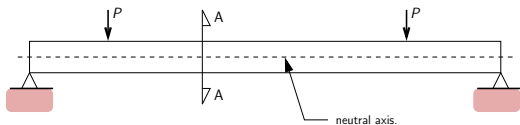
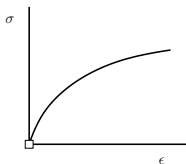
**Q3.** What is the relationship between  $V(x)$  and  $M(x)$ ? Are they independent? No!

We will see:  $V(x) = \frac{dM(x)}{dx}$ , but not always true!

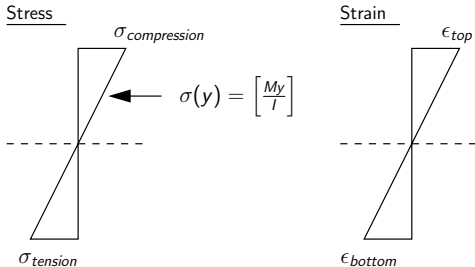
# Connection to Mechanics

# Connection to Mechanics

## Problem Setup



## Stress-Strain Relationships



# Connection to Mechanics

For design purposes we need to make sure:

$$\sigma_{tension} < \sigma_{\max \text{ tension}} \quad (1)$$

and

$$\sigma_{compression} < \sigma_{\max \text{ compression}} \quad (2)$$

Also,

$$\epsilon_{\max \text{ compression}} \leq \epsilon(y) \leq \epsilon_{\max \text{ tension}} \quad (3)$$

These constraints limit the amount of load that a beam can carry.

# Connection to Mechanics

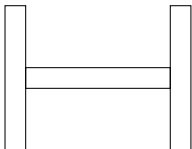
## Section-Level Behavior

From a design standpoint we can reduce  $\sigma(y)$  and  $\epsilon(y)$  by increasing the moment of inertia in

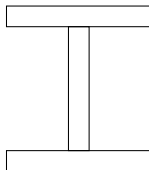
$$\sigma(y) = \left[ \frac{My}{I} \right]. \quad (4)$$

To maximise  $I$ , maximize distance of material from neutral axis.

Poor Choice of Inertia



Good Choice of Inertia

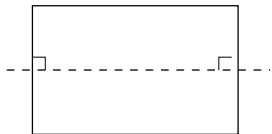


# Assumptions on Beam Displacements

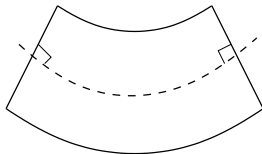
**Assumptions.** We will assume beam length / depth  $\gg 10$ .

Therefore, displacements will be dominated by flexural bending.

Undeformed Configuration



Deformed Configuration



Sections remain perpendicular to the deformed neutral axis.

This is **not the case** for **shear deformations**.

# Relationship between Shear Force and Bending Moment

# Relationship between Shear Force and Bending Moment

## Basic Questions

- Are  $V(x)$  and  $M(x)$  independent? **No!**
- Under what conditions does a dependency relationship exist?

## Strategy

- Introduce relevant mathematics.
- Extract a thin section from a beam and examine its equilibrium.
- See where the mechanics takes us!



# Mathematical Preliminaries

The Taylor series is as follows:

$$f(x+h) = \sum_{k=0}^{\infty} \frac{f^k(x)}{k!} h^k = f(x) + f'(x)h + \frac{f''(x)}{2!} h^2 + \frac{f'''(x)}{3!} h^3 + \dots \quad (5)$$

For a Taylor series approximation containing  $(n+1)$  terms

$$f(x+h) = \sum_{k=0}^{k=n} \frac{f^k(x)}{k!} h^k + O(h^{(n+1)}) \quad (6)$$

The big-O notation indicates how quickly the error will change as a function of  $h$ , e.g.,  $O(h^2) \rightarrow$  magnitude of error proportional to  $h$  squared.

# Mathematical Preliminaries

**Finite Difference Derivatives.** Truncating equation 6 after two terms gives:

$$f(x + h) = f(x) + f'(x)h + O(h^2). \quad (7)$$

A simple rearrangement of equation 7 gives:

$$\frac{dy}{dx} = \lim_{h \rightarrow 0} \left[ \frac{f(x + h) - f(x)}{h} \right]. \quad (8)$$

Similarly, we require:

$$\frac{dy}{dx} = \lim_{h \rightarrow 0} \left[ \frac{f(x) - f(x - h)}{h} \right]. \quad (9)$$

In order for the derivative to exist, equations 8 and 9 need to be the same!

# Mathematical Preliminaries

**Simple Example.** Let  $y = x^2$ .

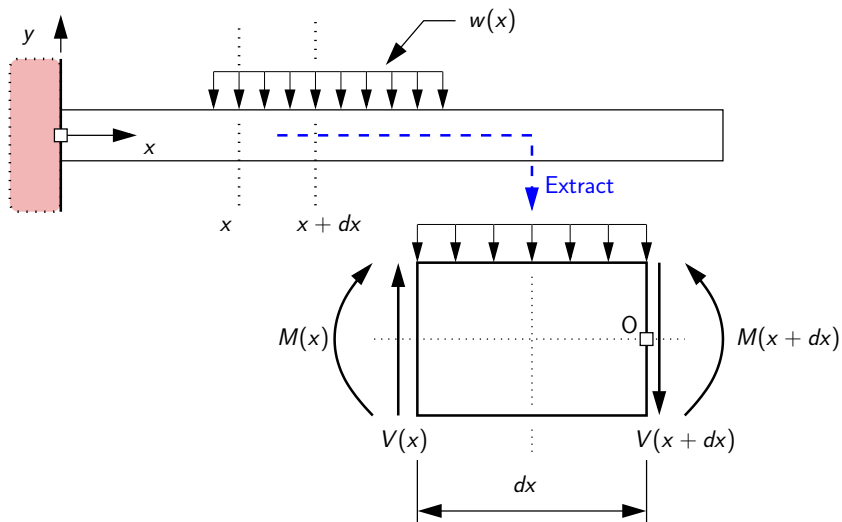
$$\frac{dy}{dx} = \lim_{h \rightarrow 0} \left[ \frac{(x+h)^2 - x^2}{h} \right] = \lim_{h \rightarrow 0} [2x + h] = 2x. \quad (10)$$

**Home Exercise.** Use first principles to find  $dy/dx$  when:

$$y(x) = (x^2 - 4x + 3)^2 \quad (11)$$

**Counter Example.**  $y(x) = |x|$  is not differentiable at  $x = 0$ .

# Test Problem for Derivation of Equations



# Derivation of Equations

**Part 1:** Equilibrium in Vertical Direction:

$$\sum F_y = 0 \rightarrow V(x) - V(x + dx) - w(x)dx = 0 \quad (12)$$

From the Taylor's series expansion:

$$V(x + dx) = V(x) + \frac{dV}{dx}dx + O(dx^2) \quad (13)$$

Plugging equation 13 into 12 and ignoring higher-order terms:

$$\sum F_y = 0 \rightarrow V(x) - \left[ V(x) + \frac{dV}{dx}dx \right] - w(x)dx = 0 \quad (14)$$

# Derivation of Equations

Hence,

$$\frac{dV}{dx} + w(x) = 0 \leftarrow \text{gradient of shear force equals } -w(x). \quad (15)$$

**Part 2:**  $\sum M_o = 0$  (anticlockwise +ve)

$$-V(x)dx - M(x) + M(x + dx) + w(x)dx \cdot \frac{dx}{2} = 0 \quad (16)$$

Note:

- The term  $w(x)dx$  is the vertical load acting on the element.
- The term  $dx/2$  is the distance from O to the centroid of loading.

# Derivation of Equations

From the Taylor Series expansion:

$$M(x + dx) = M(x) + \frac{dM}{dx} dx + O(dx^2) \quad (17)$$

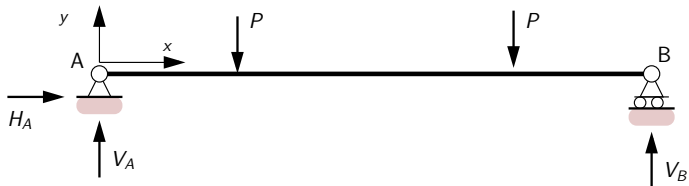
Plugging equation 17 into 16 and ignoring terms  $O(dx^2)$  and higher:

$$V(x) = \frac{dM}{dx} \leftarrow \text{shear force} = \text{gradient of bending moment.} \quad (18)$$

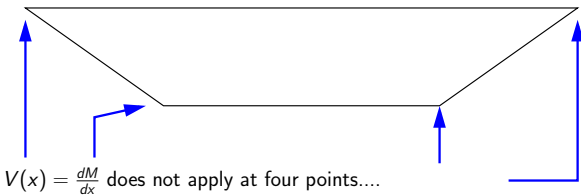
**Note.** Equation 18 only applies when the derivatives of  $M(x)$  with respect to  $x$  exist.

# Derivation of Equations

## Illustrative Example



Bending Moment Diagram



# Shear Force and Bending Moment

**Interpretation.** Consider an interval  $[a, b]$  on a beam:

$$dV = -w(x)dx \rightarrow \int_a^b dV = - \int_a^b w(x)dx = V(b) - V(a). \quad (19)$$

**Key Point:** Change in shear force between points  $a$  and  $b$  = total loading within interval.

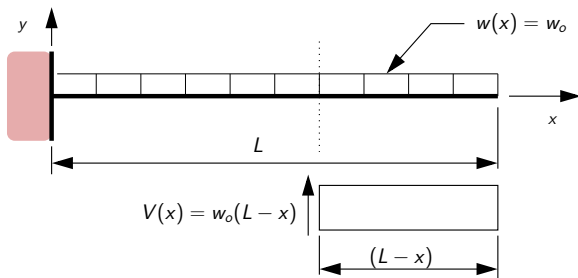
$$dM = V(x)dx \rightarrow \int_a^b dM = \int_a^b V(x)dx = M(b) - M(a). \quad (20)$$

**Key Point:** Change in moment between points  $a$  and  $b$  = area under the shear force diagram.

# Examples

# Shear Force and Bending Moment

## Example 1.



Check Shear Loading ( $a = 0$ ,  $b = L$ ):

$$V(b) - V(a) = V(L) - V(0) = -wL = -\int_0^L w_0 dx. \checkmark \quad (21)$$

# Shear Force and Bending Moment

Check Relationship between Shear and Bending Moment:

$$V(x) = \frac{dM(x)}{dx} = w_o(L - x). \quad (22)$$

For  $a = 0$  and  $b = L$  we expect:

$$\int_0^L V(x) dx = w_o \int_0^L (L - x) dx = M(L) - M(0). \quad (23)$$

For a general value  $x$ :

$$M(x) = w_o \int_x^L (L - s) ds = w_o Lx - \frac{1}{2} w_o x^2 + A. \quad (24)$$

# Shear Force and Bending Moment

Apply Boundary Conditions:

$$M(L) = 0 \rightarrow A = -\frac{1}{2}wL^2. \quad (25)$$

Hence,

$$M(x) = wLx - \frac{1}{2}wx^2 - \frac{1}{2}wL^2 = -\frac{1}{2}w(L-x)^2. \quad (26)$$

Check Moment at Boundary Conditions:

- $M(L) = wL^2 - \frac{1}{2}2wL^2 = 0. \checkmark$
- $M(0) = -\frac{1}{2}wL^2. \checkmark$

# Shear Force and Bending Moment

## Physical Interpretation

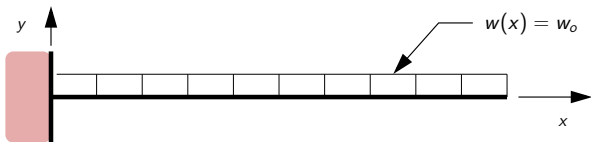
For the extracted element:

$$\sum F_y(x) = 0 \rightarrow V(x) = w_o(L - x). \quad (27)$$

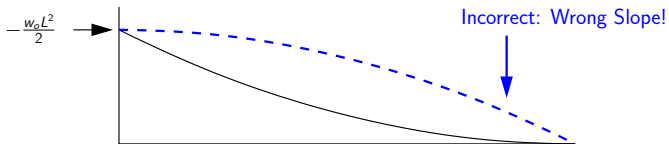
Similarly,

$$\sum M_z(x) = 0 \rightarrow M(x) = \underbrace{w_o(L - x)}_{\text{total load}} \cdot \underbrace{\frac{(L - x)}{2}}_{\text{centroid}} \quad (28)$$

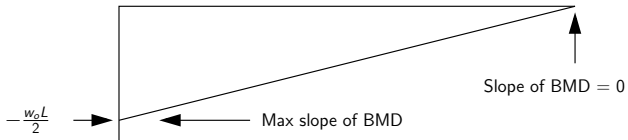
# Shear Force and Bending Moment Diagrams



Bending Moment (drawn on tension side of element):

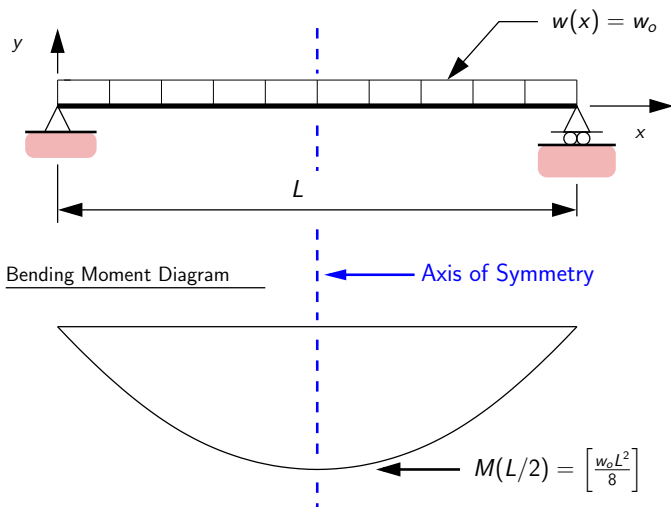


Shear Force:



# Shear Force and Bending Moment

## Example 2.





# Shear Force and Bending Moment

Equation for  $M(x)$ ?

We have:

- Axis of symmetry at  $x = L/2$ .
- $M(x)$  will have roots at  $x = 0$  and  $x = L$ .

Hence, let  $M(x) = Ax(x - L)$ , then use midpoint moment to determine  $A$ :

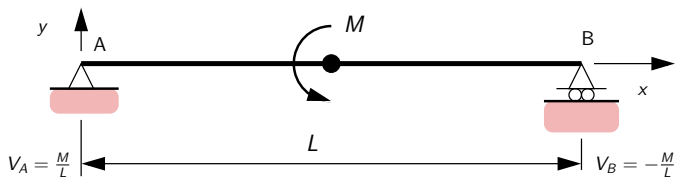
$$M(L/2) = A \frac{L}{2} \left( \frac{-L}{2} \right) \rightarrow A = -\frac{w_0}{2}. \quad (30)$$

Thus,

$$M(x) = \frac{w_0}{2} x(L - x). \quad (31)$$

# Shear Force and Bending Moment

## Example 3.



Bending Moment Diagram

